

PRIVATE PILOT COURSE

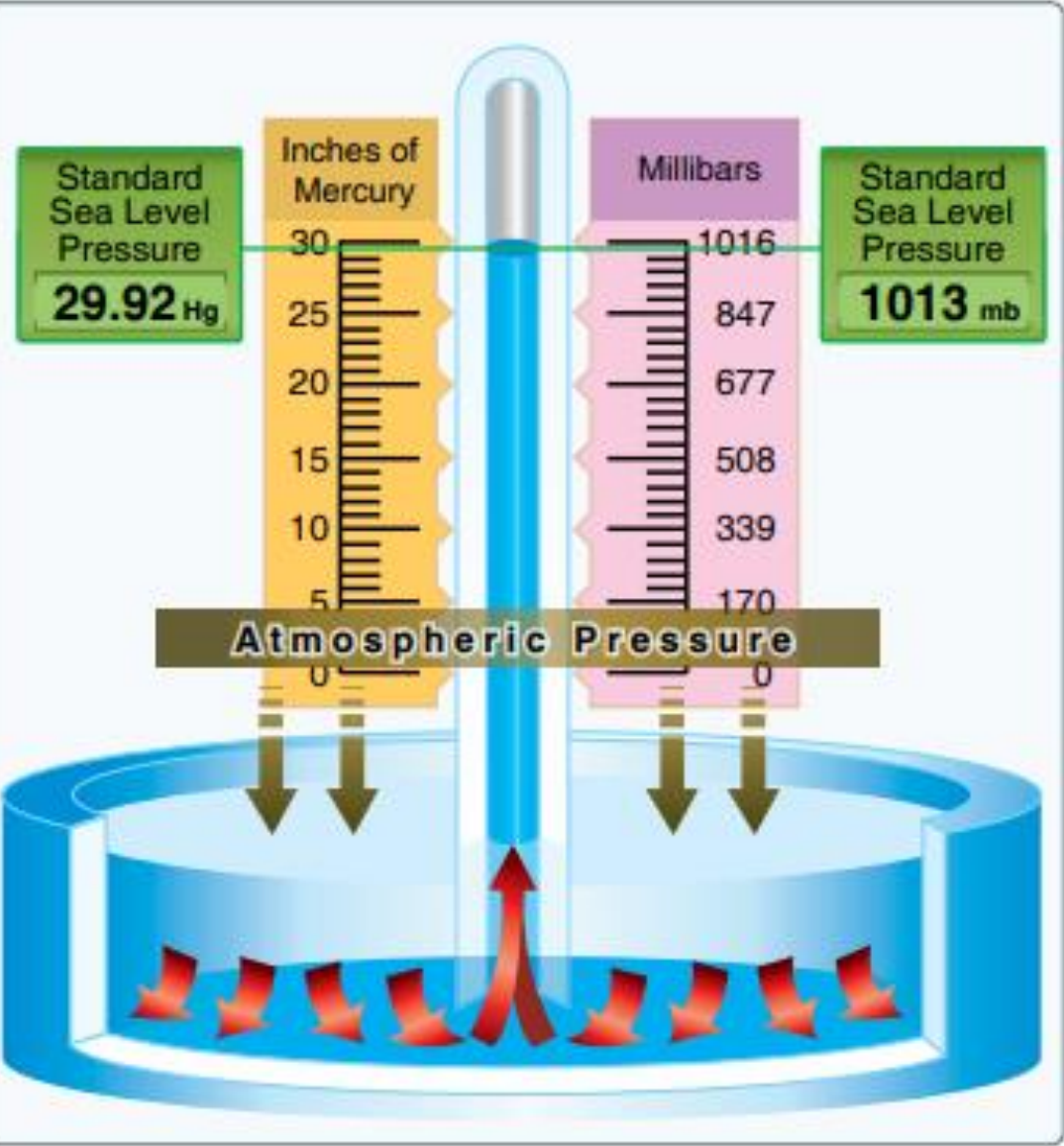
PRINCIPLES OF FLIGHT



AIR

- **Air** is made up of gases.
- **Air** has mass
- **Air** exerts pressure and has weight.
- **Air** can be compressed.
- **Air** is impacted by temperature.

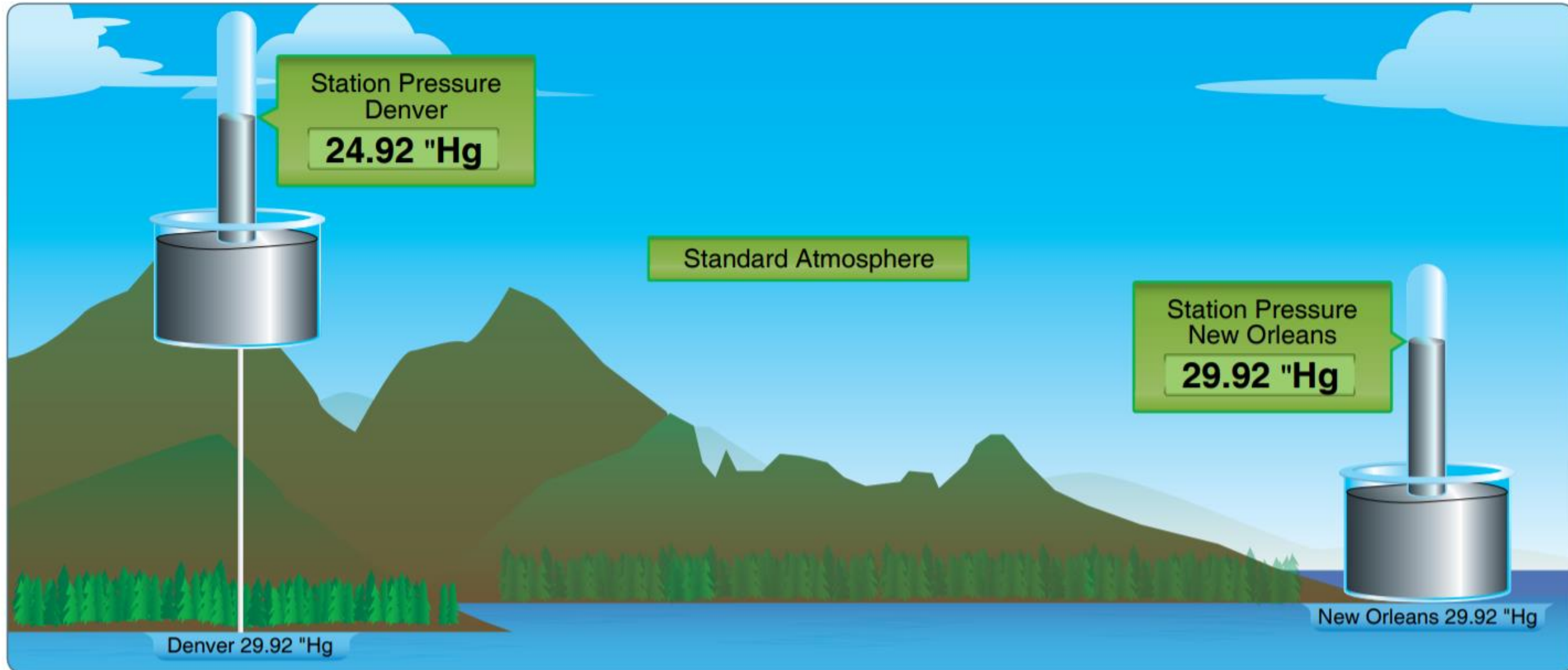
ATMOSPHERIC PRESSURE



Standard Atmosphere			
Altitude (ft)	Pressure (Hg)	Temperature	
		(°C)	(°F)
0	29.92	15.0	59.0
1,000	28.86	13.0	55.4
2,000	27.82	11.0	51.9
3,000	26.82	9.1	48.3
4,000	25.84	7.1	44.7
5,000	24.89	5.1	41.2
6,000	23.98	3.1	37.6
7,000	23.09	1.1	34.0
8,000	22.22	-0.9	30.5
9,000	21.38	-2.8	26.9
10,000	20.57	-4.8	23.3
11,000	19.79	-6.8	19.8
12,000	19.02	-8.8	16.2
13,000	18.29	-10.8	12.6
14,000	17.57	-12.7	9.1
15,000	16.88	-14.7	5.5
16,000	16.21	-16.7	1.9
17,000	15.56	-18.7	-1.6
18,000	14.94	-20.7	-5.2
19,000	14.33	-22.6	-8.8
20,000	13.74	-24.6	-12.3

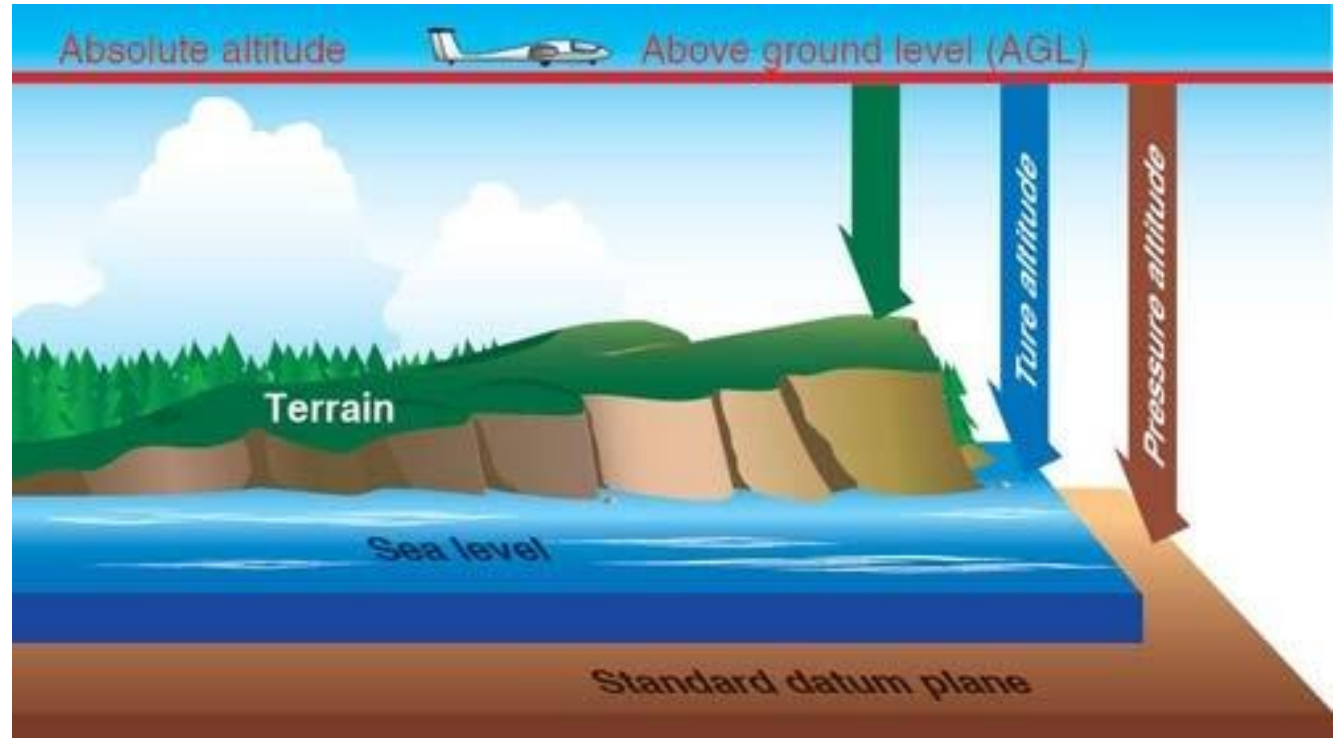


ATMOSPHERIC PRESSURE

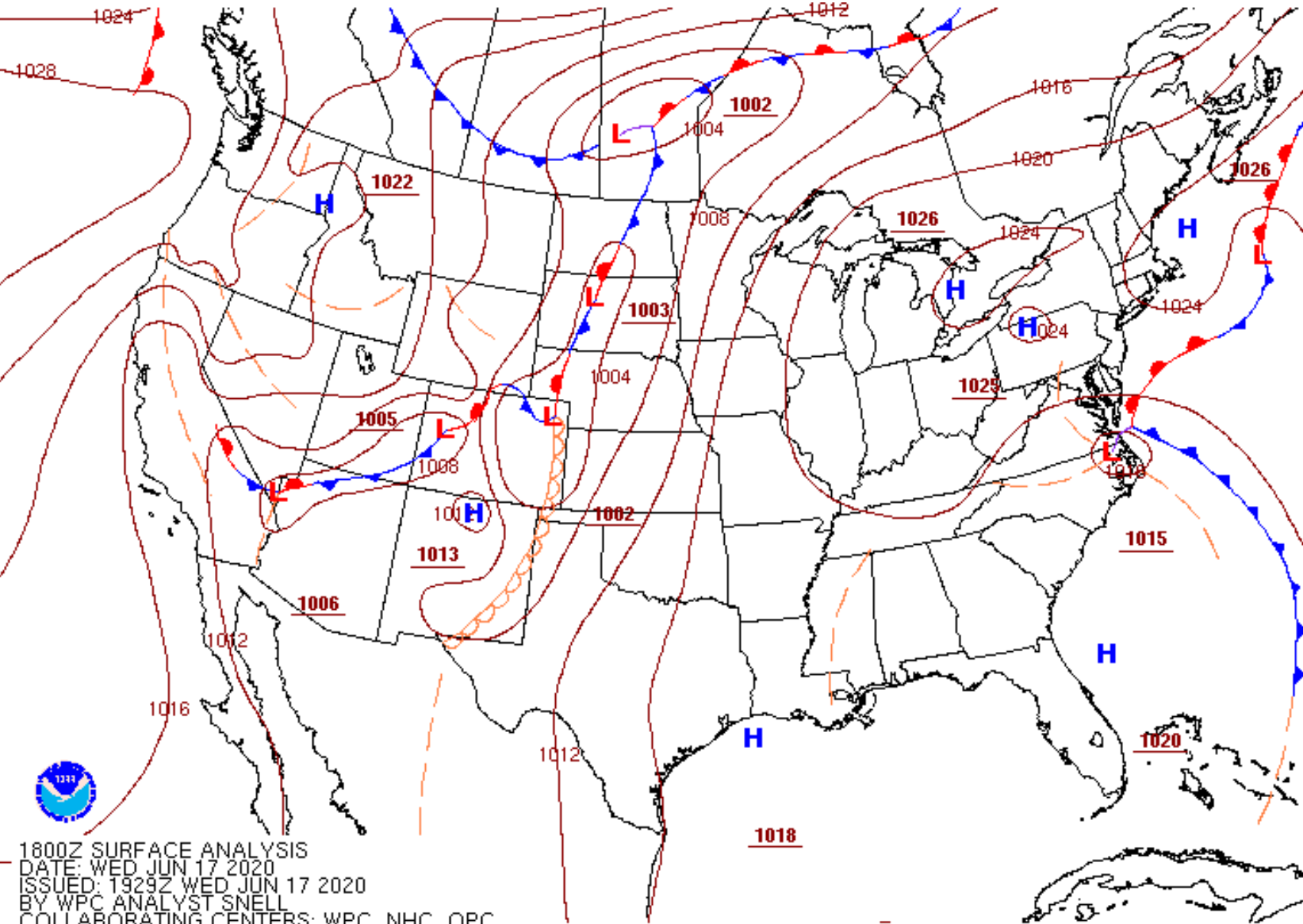


5 TYPES OF ALTITUDE

- Indicated Altitude
- Absolute Altitude
- True Altitude
- Pressure Altitude
- Density Altitude



PRESSURE



What is
standard
atmospheric
pressure?

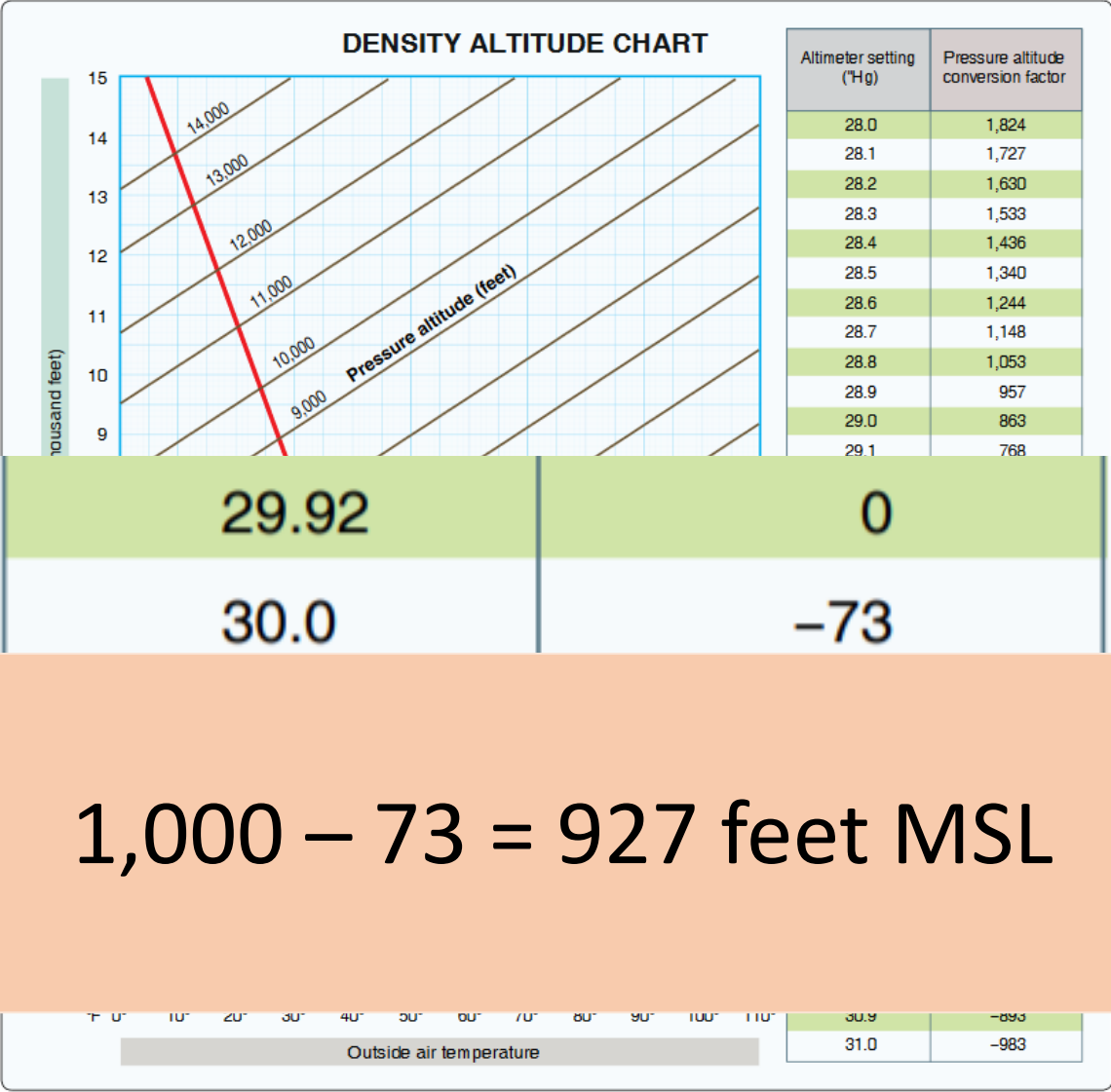
29.92"
1013 mb

PRESSURE ALTITUDE



How do we find Pressure Altitude?

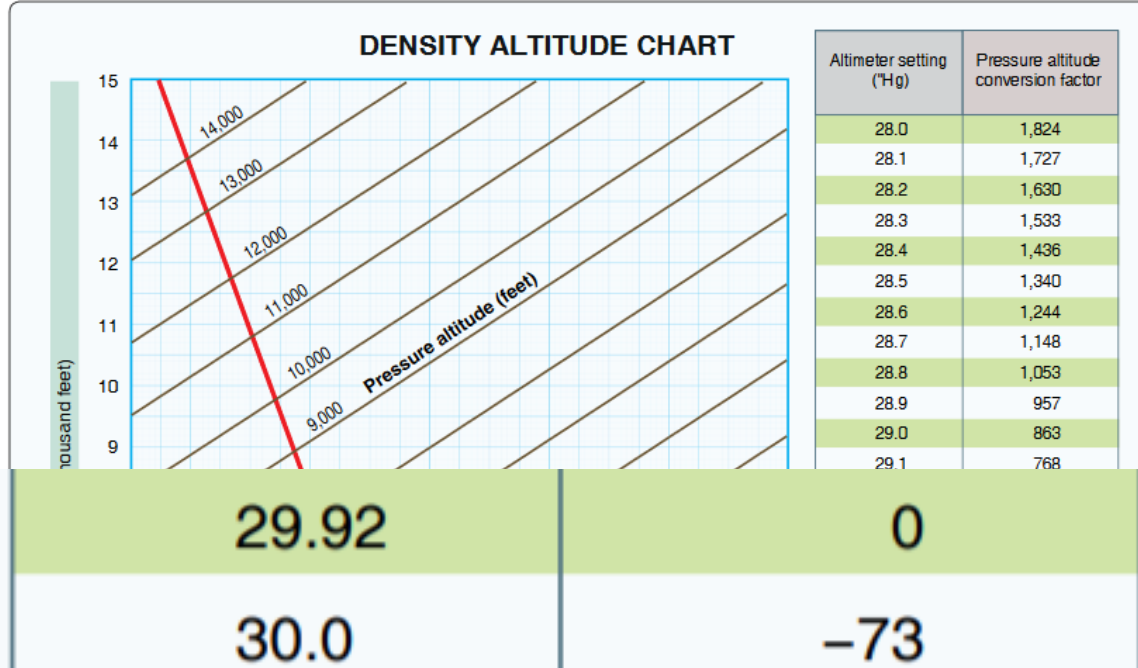
CALCULATING PRESSURE ALTITUDE



(Refer to figure 8.) Determine the pressure altitude at an airport that is 1,000 feet MSL with an altimeter setting of 30.00

$1,000 - 73 = 927 \text{ feet MSL}$

CALCULATING PRESSURE ALTITUDE



$$.01 = -9$$

$$29.96 - 29.92 = .04$$

$$4 \times -9 = -36$$

$$3,563 - 36 = 3,527 \text{ feet MSL}$$

(Refer to figure 8.) Determine the pressure altitude at an airport that is 3,563 feet MSL with an altimeter setting of 29.96.

- a) 3,556 feet MSL
- b) 3,527 feet MSL
- c) ~~3,563 feet MSL~~

CALCULATING PRESSURE ALTITUDE

Rule of Thumb
 $1" = 1,000 \text{ feet}$

(Refer to figure 8.) Determine the pressure altitude at an airport that is 3,563 feet MSL with an altimeter setting of 29.96.

MATH

$$29.96 - 29.92 = .04$$


$$.04 \times 1,000 = 40 \text{ feet}$$

$$3,563 - 40 = \mathbf{3,523 \text{ feet}}$$

- a) 3,556 feet MSL
- b) 3,527 feet MSL
- c) 3,639 feet MSL

CALCULATING PRESSURE ALTITUDE

Altimeter setting ("Hg)	Pressure altitude conversion factor
28.0	1,824
28.1	1,727
28.2	1,630



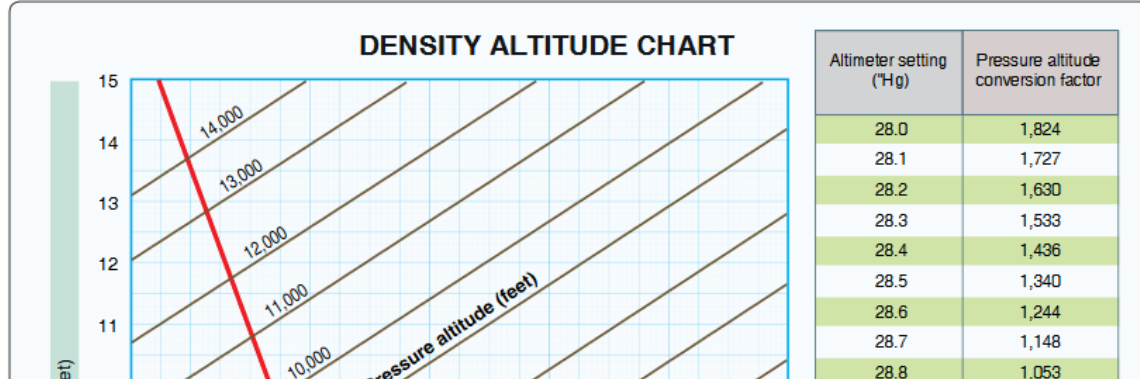
Determine the pressure altitude with an indicated altitude of 1,380 feet MSL with an altimeter setting of 28.22 at standard temperature.

- a) 3,010 feet MSL.
- b) 2,991 feet MSL.
- c) 2,913 feet MSL.

$$1,630 + 1,380 = 3,010 \text{ feet MSL}$$

°C -18°	-12°	-7°	-1°	4°	10°	16°	21°	27°	32°	38°	43°	30.8	-803
°F 0°	10°	20°	30°	40°	50°	60°	70°	80°	90°	100°	110°	30.9	-893
Outside air temperature												31.0	-983

CALCULATING PRESSURE ALTITUDE



29.92

0

30.0

-73

$$.01 = -9$$

$$29.97 - 29.92 = .05$$

$$5 \times -9 = -45$$

$$1,386 - 45 = 1,341 \text{ feet MSL}$$

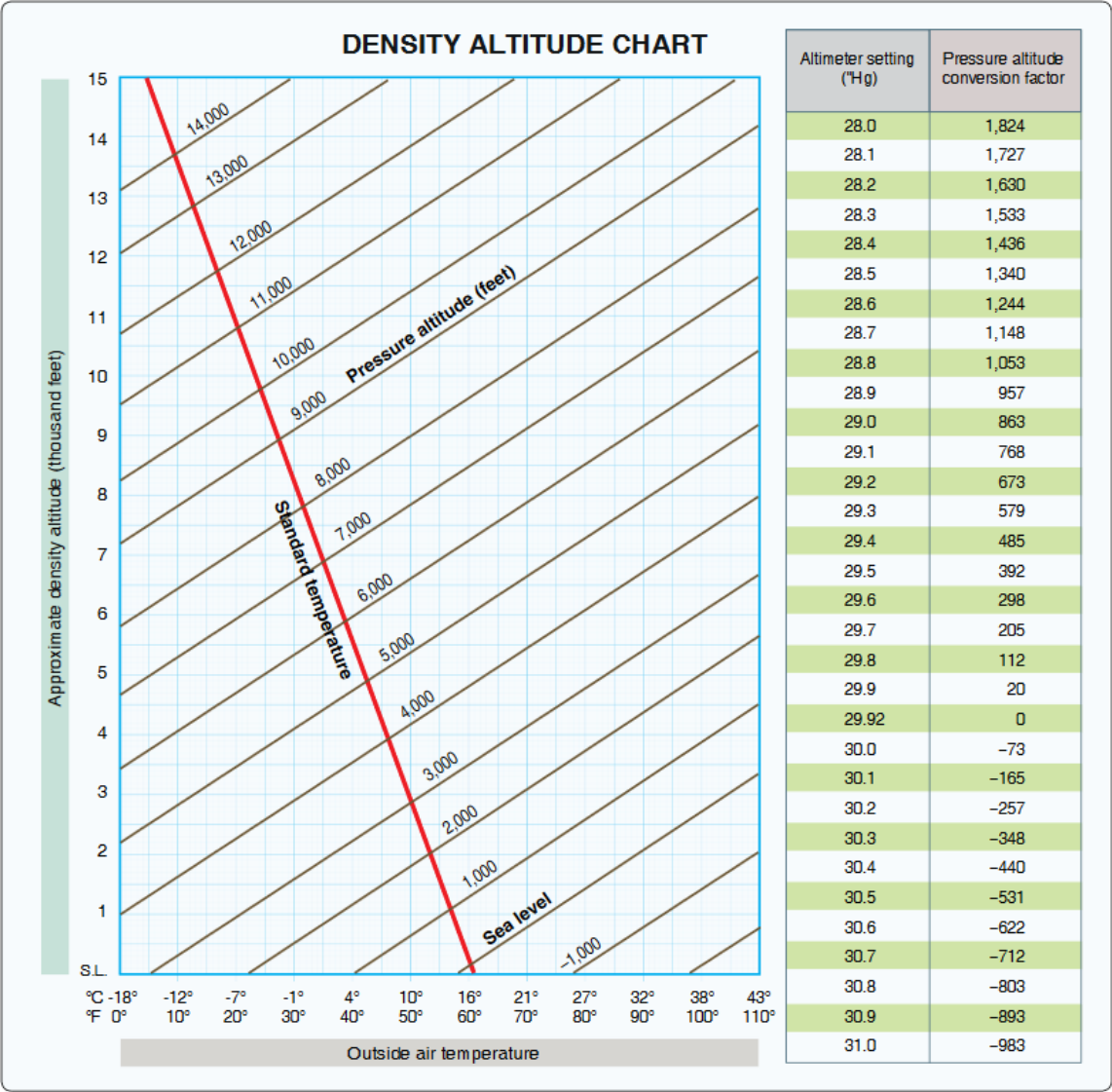
(Refer to figure 8.) Determine the pressure altitude at an airport that is 1,386 feet MSL with an altimeter setting of 29.97.

- a) 1,451 feet MSL.
- b) 1,341 feet MSL.
- c) 1,562 feet MSL.

DENSITY ALTITUDE



CALCULATING DENSITY ALTITUDE

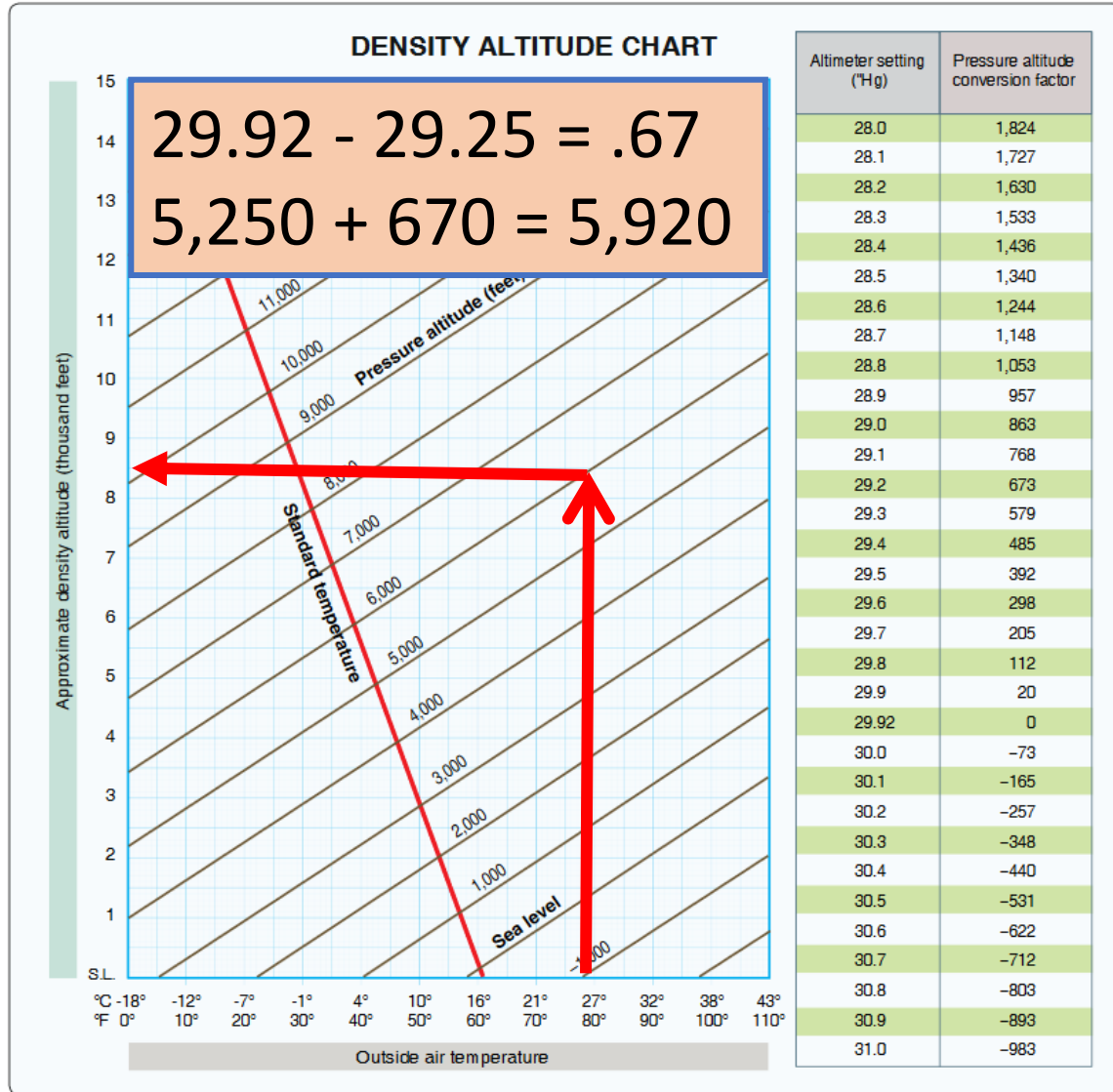




32° C
5000 FT.

TheFinnerPointe.net™

CALCULATING DENSITY ALTITUDE



(Refer to figure 8.) Determine the density altitude for these conditions:

Altimeter setting 29.25

Runway temperature +81 °F

Airport elevation 5,250 ft MSL


- a) 8,500 feet MSL
- b) 5,877 feet MSL
- c) 4,600 feet MSL.

HUMIDITY

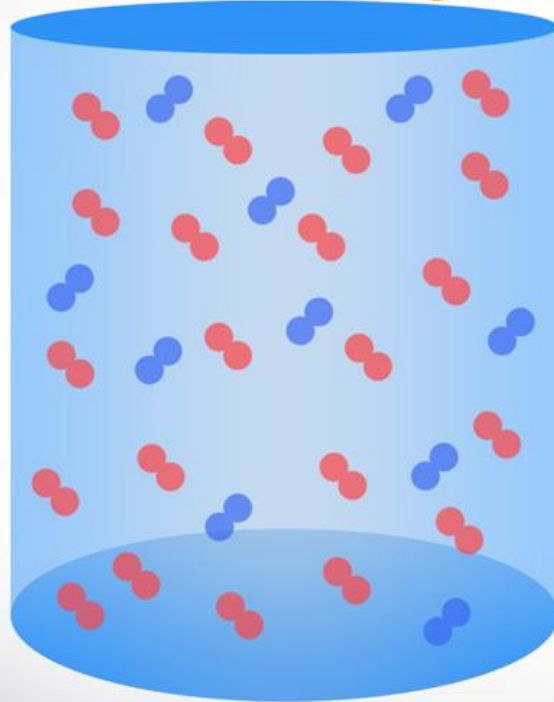
Humidity

 = Nitrogen (N₂)

 = Oxygen (O₂)

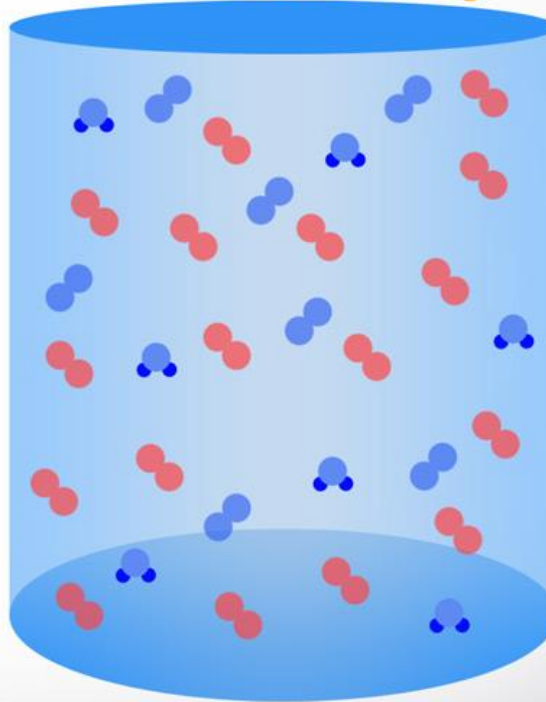
 = Water Vapor (H₂O)

0% Humidity



Mass = 440

100% Humidity



Mass = 420

[boldmethod](#) ▶



DENSITY ALTITUDE



Newton's Laws of Motion

Newton's laws of motion in physics

LAW #1

A body at rest will remain at rest, and a body in motion will remain in motion unless it is acted upon by an external force.

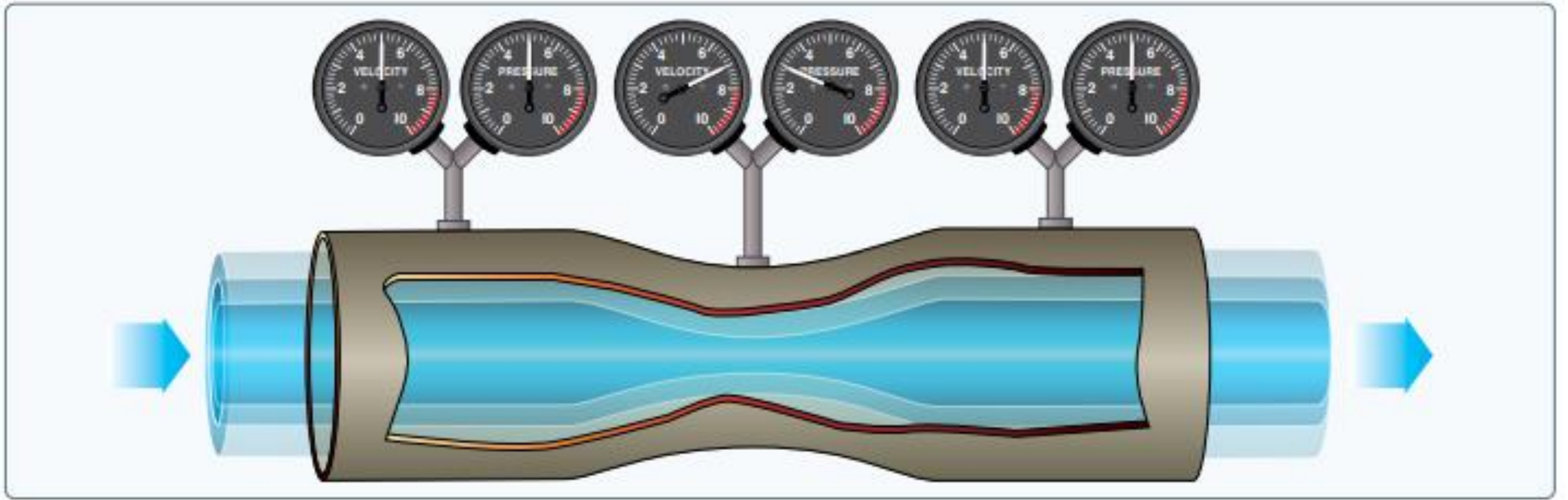
LAW #2

The force acting on an object is equal to the mass of that object times its acceleration, $F = ma$.

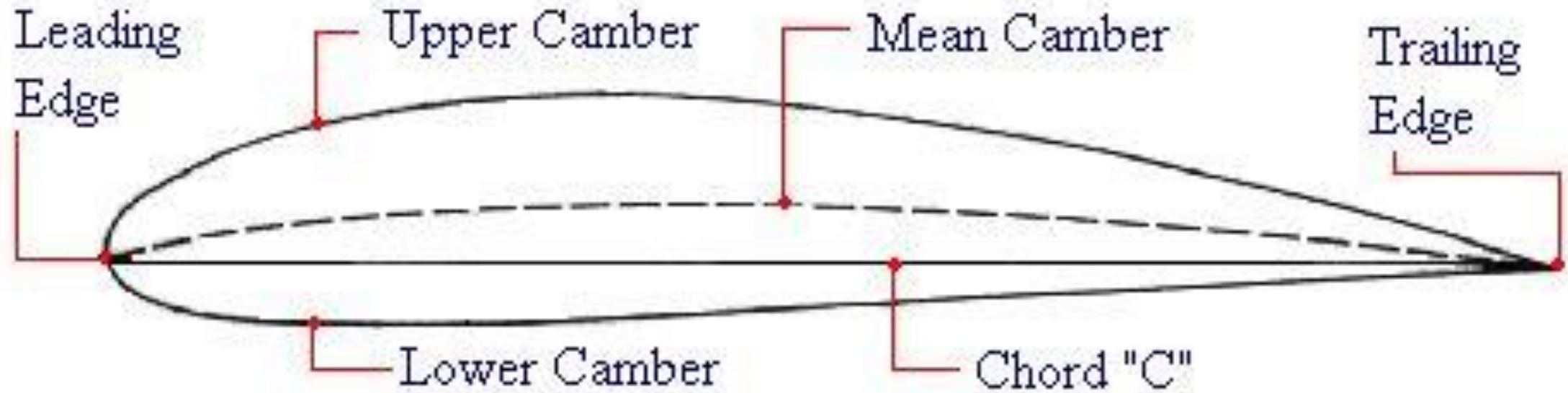
LAW #3

For every action, there is an equal and opposite reaction.

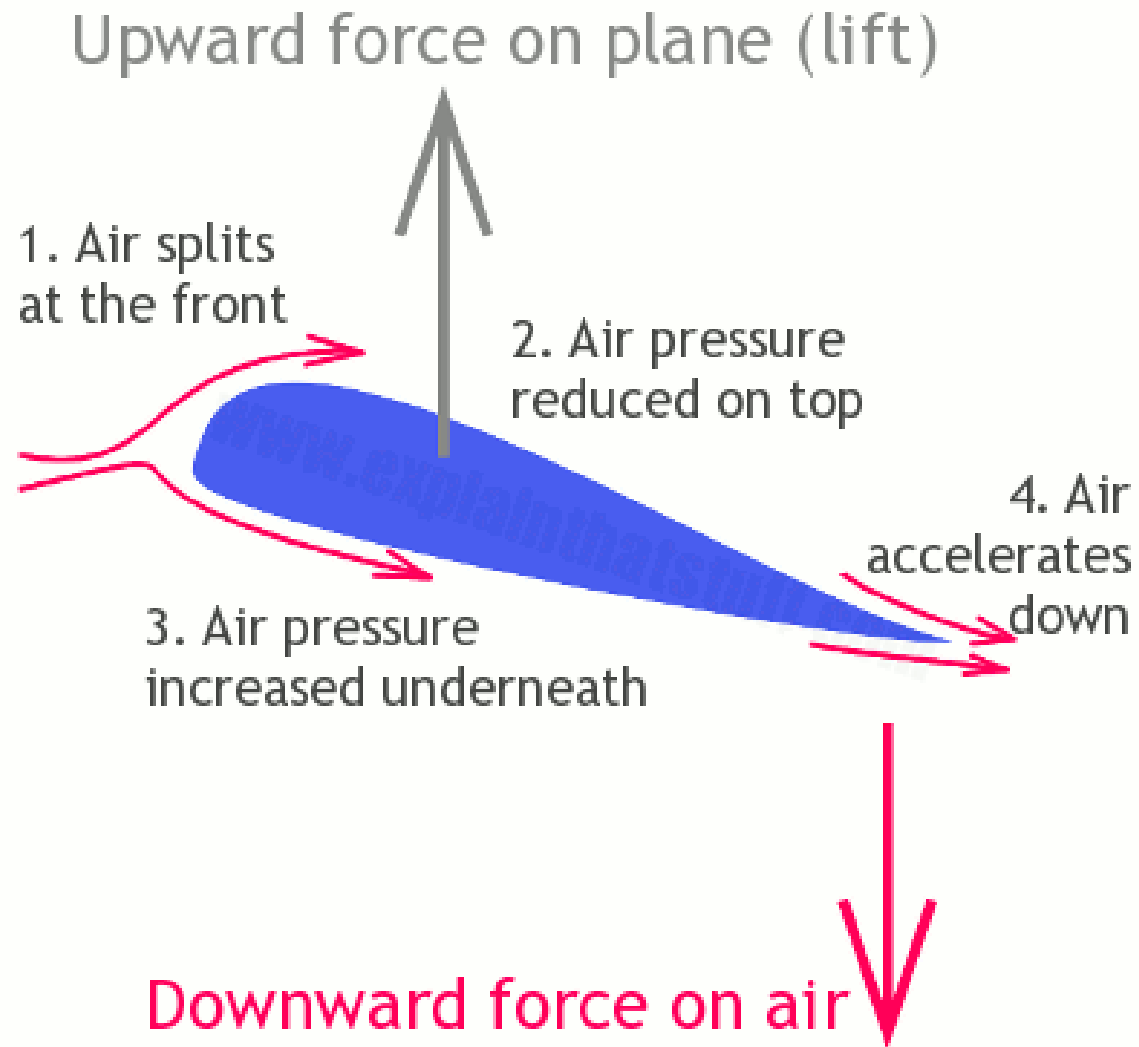
Bernoulli's Principle



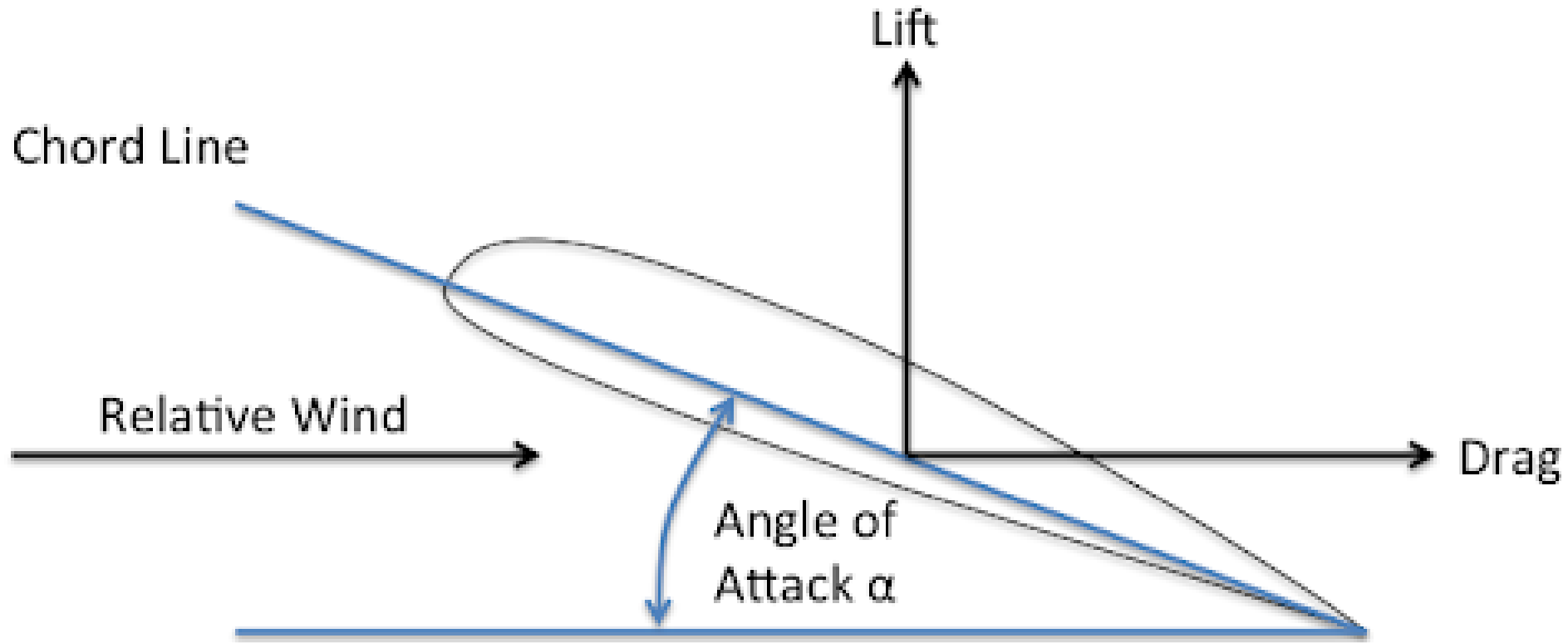
Airfoil Design



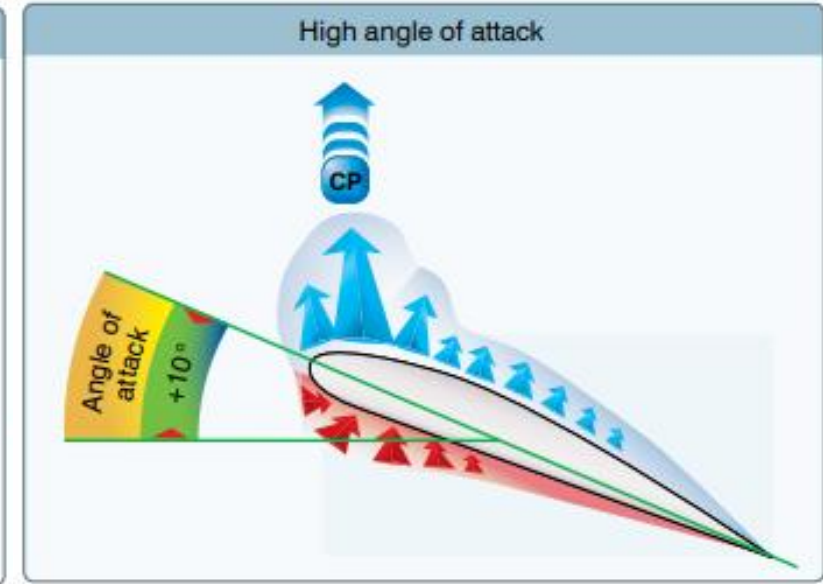
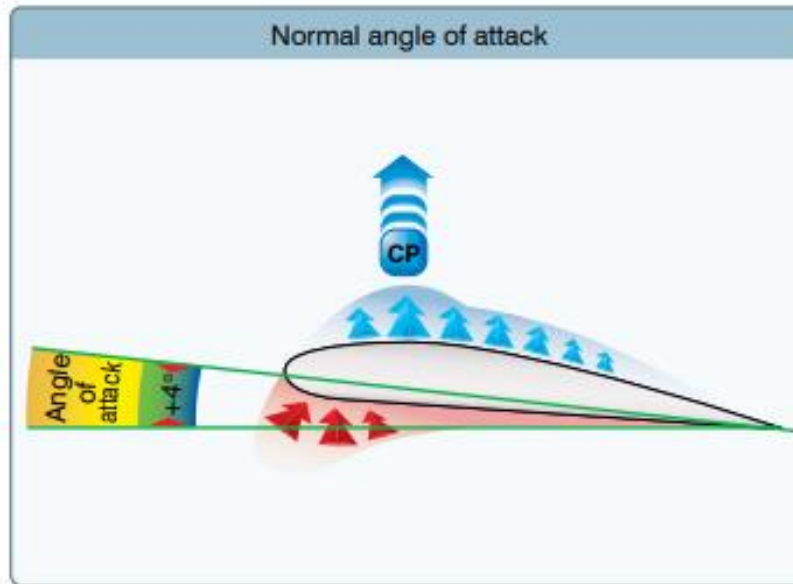
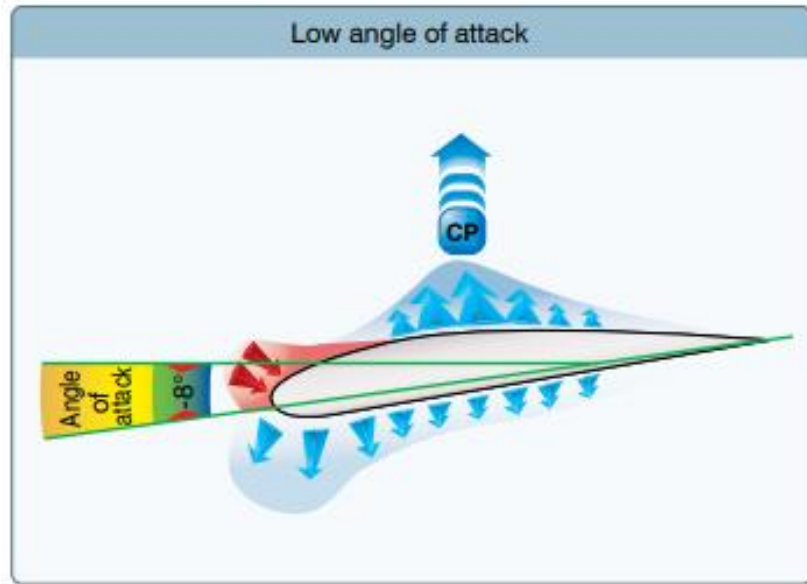
Bernoulli's Principle



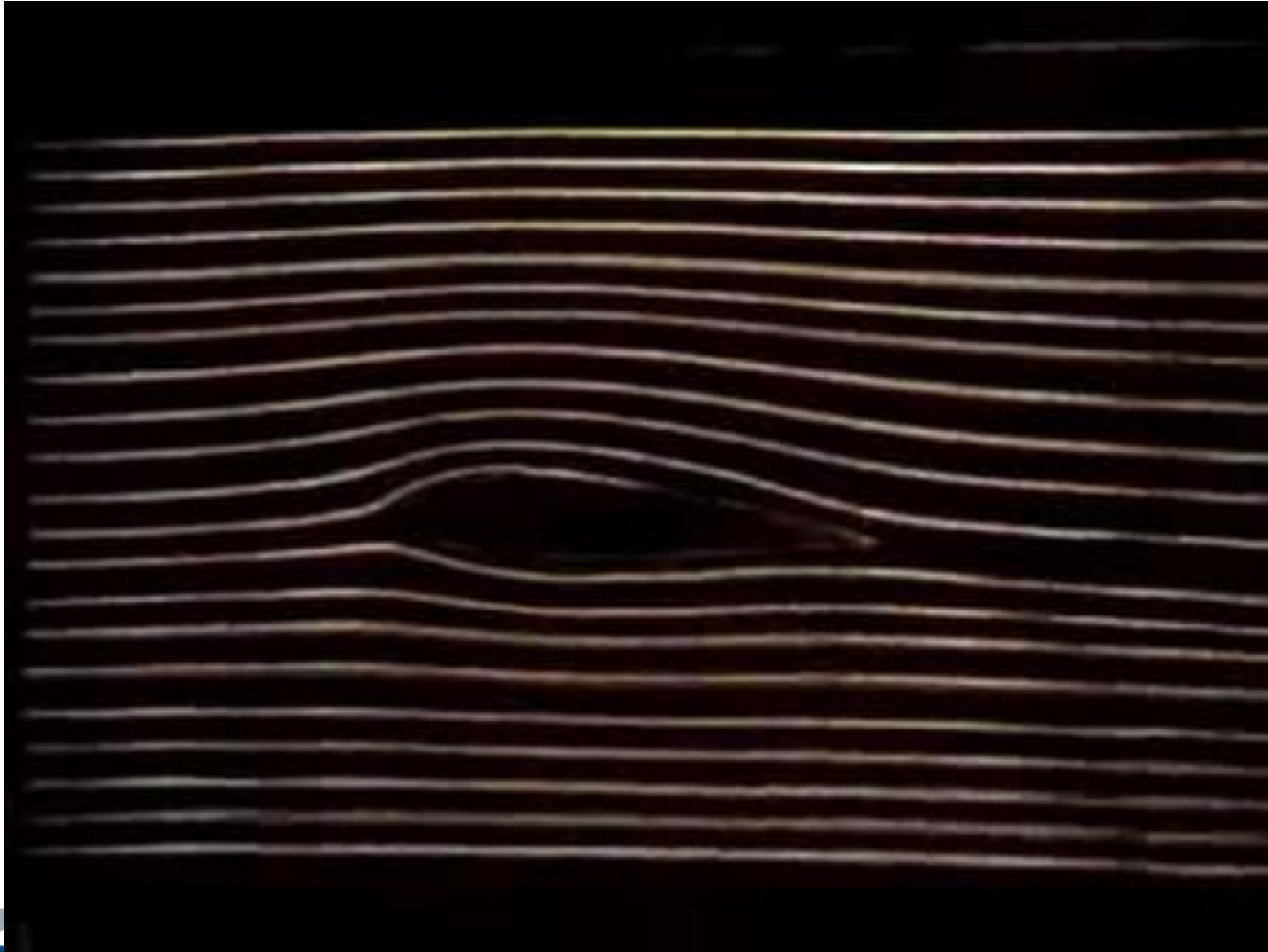
Angle of Attack



Angle of Attack – Pressure Distribution



Wind Tunnel Demo



SHAPE

Early airfoil



Later airfoil



Clark 'Y' airfoil
(Subsonic)



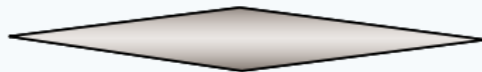
Laminar flow airfoil
(Subsonic)



Circular arc airfoil
(Supersonic)



Double wedge airfoil
(Supersonic)



Wingtip Vortices

